

FUNDAMENTALS OF SOLID PROPELLANT COMBUSTION

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 - Thanks to Dr. John C. Yang and Dr. William L. Grosshandler of NIST for their invitation for my participation in this workshop.

COMMENTS ON WORKSHOP TOPIC

- It is very exciting to see that solid propellants are being considered for gas generator application in fire extinguishment.
- Great Engineering Challenge!!

GENERAL BACKGROUND OF SOLID PROPELLANTS

- (1) SOLID STATE SUBSTANCES WHICH CONTAIN BOTH OXIDIZERS AND FUEL INGREDIENTS
- (2) ABLE TO BURN IN ABSENCE OF AMBIENT AIR OR OXIDIZERS
- (3) NORMALLY USED TO GENERATE HIGH-TEMPERATURE COMBUSTION PRODUCTS FOR PROPULSION PURPOSES
- (4) CLASSIFIED INTO TWO DIFFERENT TYPES (HOMOGENEOUS AND HETEROGENEOUS) BASED ON DIFFERENCES IN THEIR PHYSICAL STRUCTURE

CLASSES OF PROPELLANTS

- Homogeneous
 - Uniform physical structure.
 - Fuel and oxidizer are chemically bonded together.
 - Major constituents are nitrocellulose (NC) and nitroglycerine (NG).
 - Also referred to as double-base propellants.
- Heterogeneous
 - Non-uniform physical structure.
 - Polymeric fuel binder and crystalline oxidizers.
 - Also referred to as composite propellants.

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Edited by

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It provides an excellent tutorial introduction of the performance of solid-propellant rocket motors and a general survey of solid-propellant combustion characteristics.

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by K. Kishore and V. Gayathri

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presents the basic chemistry of ignition and combustion of AP-based propellants as well as the roles of oxidizers, binder, catalysts, and ambient conditions in ignition and combustion.

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describes detailed kinetics of NC, HMX, and RDX decomposition catalysis of nitramine propellants, as well as flame-zone chemistry

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Covers detailed description of various types of ignition experiments and theories, ignition phenomena and criteria.

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presents detailed discussions on flame spreading over the surface of solid propellant and into the defects of propellant grains.

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presents useful information on the flame structure and combustion mechanism of homogeneous propellants, mechanism of super rates and Plateau or Mesa effect.

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gives detailed treatment of various models developed for heterogeneous propellants

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Covers various theoretical models which are grouped into different categories based upon the method of approach. Presents erosive burning data, observed phenomena, and basic mechanisms.

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provides basic mechanisms, theoretical models, experimental observations, and effects of physicochemical parameters on transient burning rates

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Presents extinction theories and experiments. Covers dynamic extinction of fast depressurization, deradiation, and various quenching techniques.

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discusses experimentally observed combustion instabilities in rocket motors. Covers techniques for the measurement of combustion response functions various modes of instabilities, effect of propellant characteristics on combustion instabilities

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deals with theoretical analysis of combustion instability including linear analysis of wave motion, acoustic amplification and damping, non-acoustic instability nonlinear analysis and predictive capabilities of various theories.

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contains detailed discussions of the chemical origins of smoke secondary smoke formation and its modeling, homogeneous and heterogeneous nucleation of smoke, and various methods of reducing smoke of propellant products.